

AIRPLANE MODELS
RP-40D, P-40E & E1

SPECIFIC ENGINE FLIGHT CHART

ENGINE MODELS
V-1710-39

(WITH STREAMLINE MANIFOLDS)

CONDITION	FUEL PRESSURE (LB./SQ. IN.)	OIL PRESSURE (LB./SQ. IN.)	OIL TEMP.		COOLANT TEMP.		MAX. PERMISSIBLE DIVING RPM: 3120
			°C	°F	°C	°F	
DESIRED	12-16	60-70	60-80	140-176	105-118	220-240	
MAXIMUM	16	85	95	203	125	257	
MINIMUM	12	55			85	185	
IDLING	9	15					

CONDITION	ALLOWABLE OIL CONSUMPTION
MAX. CONT.	.12-3... U.S.QT./HR. 22-2... IMP.PT./HR
MAX. CRUISE	.19... U.S.QT./HR. 19-6... IMP.PT./HR
MIN. SPECIFIC	.5-7... U.S.QT./HR. 8-11... IMP.PT./HR

OIL GRADE: (S) 1120 (W) 1100

SUPERCHARGER TYPE: SINGLE STAGE

FUEL GRADE: 8-1 + 1.0

OCTANE *

OPERATING CONDITION	RPM	MANIFOLD PRESSURE (BOOST)	HORSE-POWER	CRITICAL ALTITUDE		BLOWER	USE LOW BLOWER BELOW:	MIXTURE CONTROL POSITION	FUEL FLOW (GAL./HR./ENG.)		MAXIMUM CYL. TEMP.		MAXIMUM DURATION (MINUTES)
				WITH RAM	NO RAM				U.S.	IMP.	°C	°F	
TAKE-OFF	3000	45.5	1150	Sea Level				Auto Rich	125	106			5
WAR EMERGENCY	3000	56	1470	Sea Level				Auto Rich	163	136			5
MILITARY	3000	44.2	1150	11,000				Auto Rich	132	110			15
MAXIMUM CONTINUOUS	2600	37.8	1000	11,000				Auto Rich	105	88			
MAXIMUM CRUISE	2280	31.0	750	11,000				Auto Rich	65	54			
MINIMUM SPECIFIC CONSUMPTION	1950	22.5	395	Sea Level				Auto Lean	30	25			
	1950	22.5	420	5,000				Auto Lean	32	27			
	1950	22.5	450	10,000				Auto Lean	34	28			
	1950	22.5	475	15,000				Auto Lean	36	30			
	1950	F.T.	460	20,000				Auto Lean	35	29			

REMARKS: * Fuel Specification AN-VV-F-781 (Amend. 5)

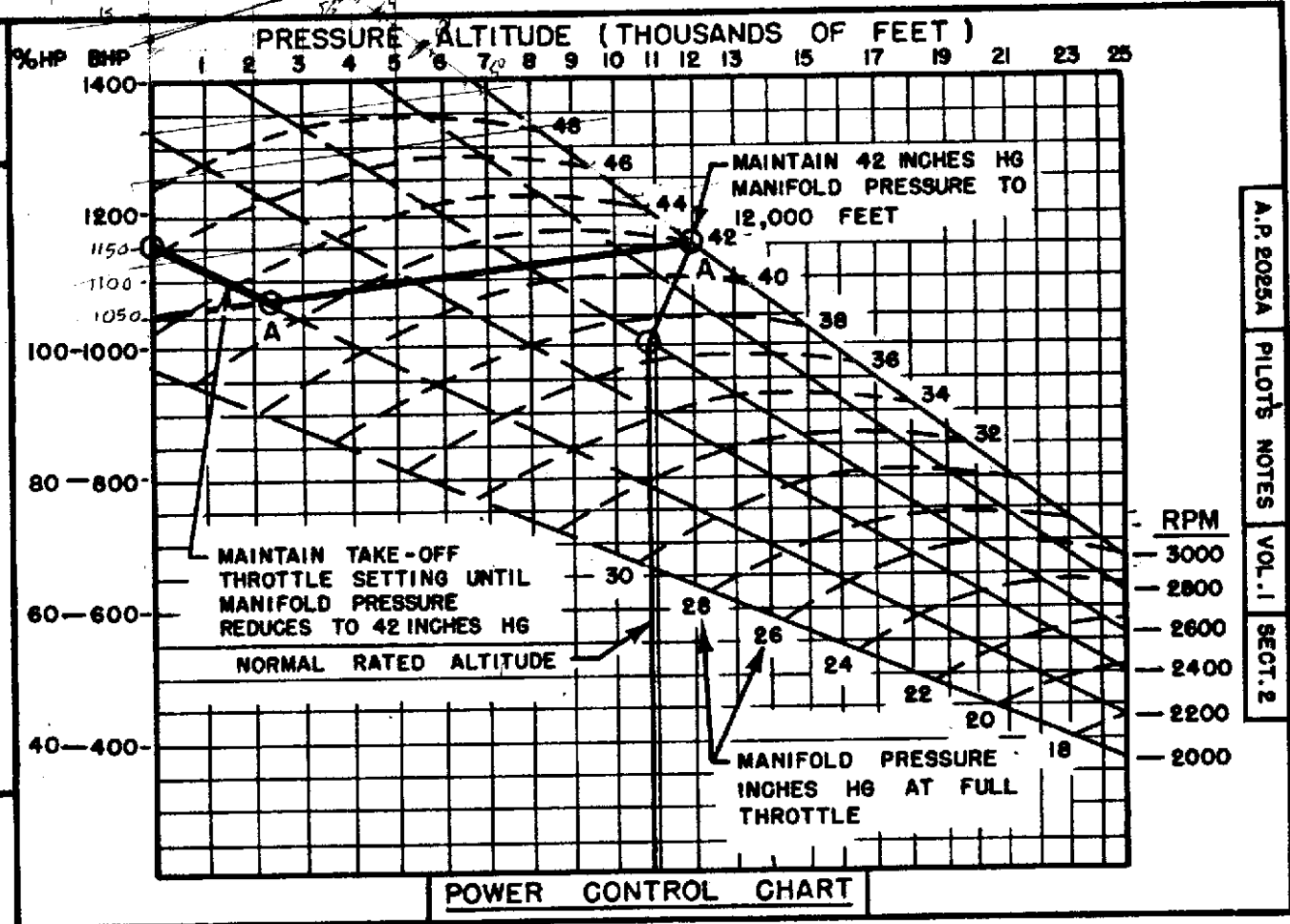
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2995 BHP
3000 BHP

Est. 1820 hp @
3000 RPM

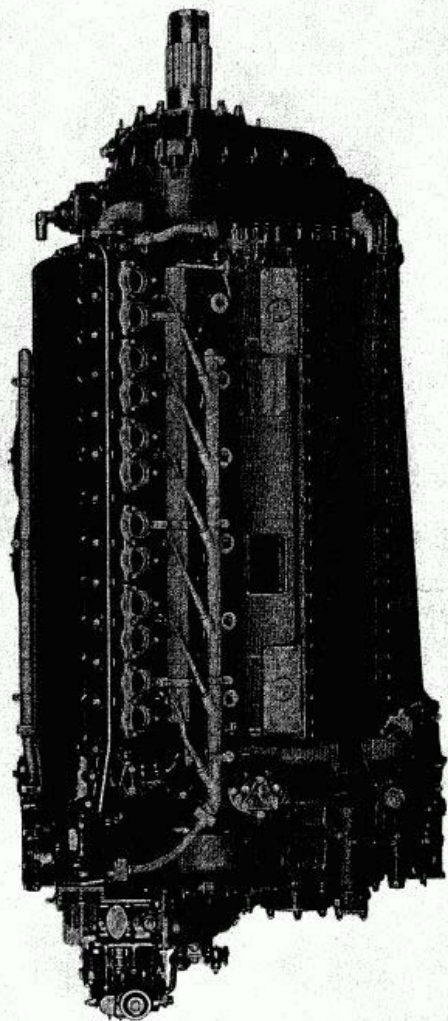
FIG. 1

POWER CONTROL CHART

FIG. 1



A.P. 2025A PILOTS NOTES VOL. I SECT. 2



Allison V-1710-F

Allison V-1710-F

Model	V-1710-F3R (39) (military).	
Type	12 cylinders, vee 60 degrees, ethylene glycol cooled, geared drive, supercharged, 4-cycle.	
Construction	2-piece aluminum alloy crankcase. 2 cylinder blocks each consisting of 6 steel cylinder barrels shrunk in a 1-piece aluminum alloy head with an aluminum alloy coolant jacket attached to the head and to each of the 6 barrels. Each head-cylinder-jacket assembly attached to crankcase by 14 long stud bolts extending through head. 2 inlet valves and 2 exhaust valves (sodium cooled) per cylinder actuated by overhead camshaft. 6-throw 1-piece counterbalanced crankshaft supported in 6 plain bearings and 1 roller bearing. Spur reduction gear, ratio 0.50:1. Tractor propeller.	
Supercharger	Gear-driven 1-speed supercharger, ratio 8.8:1.	
Carburation	1 Bendix-Stromberg PD-12K2 2-barrel injection type downdraft carburetor with automatic mixture control.	
Ignition	1 Bendix-Scintilla DF dual magneto and 2 12-point distributors.	
Starter	Eclipse hand and/or electric starter.	
Bore	5.50 in.	140 mm
Stroke	6.00 in.	152 mm
Displacement	1,710 cu.in.	28.0 l
Compression ratio	6.65:1	6.65:1
Width	29.3 in.	0.745 m
Height	36.5 in.	0.927 m
Length	35.6 in.	2.174 m
Frontal area	6.1 sq.ft.	0.57 m ²
Weight	1,310 lb.	594 kg
Weight/horsepower	1.14 lb.	0.51 kg
Fuel consumption (cr.)	0.45 lb./h.p./hr.	205 g/hp/hr
Oil consumption (cr.)	0.020 lb./h.p./hr.	9 g/hp/hr
Gasoline rating	100 octane	100 octane
Output/displacement	0.67 h.p./cu.in.	41.1 hp/l
Output/piston area	4.03 h.p./sq.in.	0.63 hp/cm ²
Piston speed (max.)	3,000 ft./min.	15.2 m/sec
B.m.e.p. (max.)	177 lb./sq.in.	12.4 kg/cm ²
Output (take-off)	1,150 h.p./3,000 r.p.m./45.5 in. (1 156 mm) Hg. boost	
Output (rated)	1,150 h.p./3,000 r.p.m./12,000 ft. (3 650 m)	
Output (cruising)	750 h.p./2,280 r.p.m./10,800 ft. (3 300 m)	
The latest model of this engine develops 1,325 h.p. for take-off.		
V-1710-C15 (33):	1,040 h.p./3,000 r.p.m./take-off; 1,090 h.p./3,000 r.p.m./13,200 ft. (4 000 m) rated output. Reduction gear ratio 0.50:1. 1-speed supercharger, ratio 8.8:1. 100-octane gasoline. Tractor propeller.	
V-1710-D2 (23):	Same performance as V-1710-C15 (33). Geared drive. 1-speed supercharger. 100-octane gasoline. Equipped with extension drive shaft for pusher propeller.	
V-1710-F2R (27):	1,150 h.p./3,000 r.p.m./take-off and rated output at sea level. Geared drive. 1-speed supercharger. 100-octane gasoline. Tractor propeller.	
V-1710-F2L (29):	Same as V1710-F2R (27). Opposite propeller rotation.	



THIS PAGE IS REPRODUCED FROM A BADLY FADED OR ILLEGIBLE SOURCE.
SCANNING THIS ITEM AT A HIGHER RESOLUTION WILL NOT IMPROVE ITS LEGIBILITY.

-5-

TABLE I.

POWER PLANT INSTALLATION.Engine Details.

(Extracted chiefly from U.S. Army Air Corps Technical Order No. 02-3AB-1A)

Manufacturer: Allison Division of General Motors Corporation,
Indianapolis, Indiana.

Engine name: Air Corps Type No. Allison V-1710-39 (Maker's Type
No. V-1710-F3R).

Serial Nos.: No. 1-10019, No. 2-3400.

Type: 12 cylinder liquid cooled Vee - single stage super-
charger.

Weight: 1310 lbs. dry (maker's figure).

Specification:

Bore	5.50"
Stroke	6.00"
Capacity - cubic ins.	1710
Compression Ratio	6.65:1
Supercharger Ratio	8.80:1
Airscrew gear Ratio	0.50:1

<u>Power Ratings:</u>	<u>B.H.P.</u>	<u>R.P.M.</u>	<u>Man. Press. ins.Hg.</u>	<u>Mixture</u>	<u>Height Ft.</u>	<u>Fuel Cons. G.P.H.</u>
Take-off	1150	2800	46.2	A.R.	S.L.	125
Military (5 min.)	1200	3000	43.9	A.R.	11,200	125
Rated (Max. cont.)	1000	2600	38.7	A.R.	10,200	103
75% Cruise	750	2280	31.6	A.R.	10,200	61
67% "	670	2280	28.9	A.R.	10,200	55
60% "	600	2190	27.4	A.R. A.L. A.B.	10,200	49 50 45
Max. Economy Cruise	375 430 475 520 420	1950 1950 1950 1950 1950	24.6 24.6 24.6 24.6 R.T.	A.L. A.L. A.L. A.L. A.L.	S.L. 5,000 10,000 15,000 20,000	27.5 31.6 34.9 38.2 31.6

Diving Rating: 3120 R.P.M. (30 secs.)

Recommended
Fuel: 100 octane.

Carburettor: PD-12K2 pressure type with automatic mixture control.



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Lubrication System.

Desired inlet oil temperature	60-80°C.
Maximum " " "	95°C.
Desired oil pressure	70-70 lbs./in. ²
Maximum " " "	65 " "
Minimum " " "	55 " "
Idling " " "	15 " "
Maximum allowable oil consumption -	
At rated power	13.3 quarts/hour
" cruising power	10.0 " "

Cooling System

Desired coolant temperature	105-115°C.
Maximum " " "	125°C.
Minimum " " "	85°C.

KITTYHAWK -

POWER TABLE - ALLISON V1710-39 ENGINE

100 OCTANE FUEL.

Operating Condition	B. H. P.	R. P. M.	Man press.	Altitude Density	Mixture Control	Fuel Flow G. P. H.	Performance M. P. H.		Remarks
							Normal	Belly Tank	
Take off	1150	2800	46.2	S/Level	A. R.	104	-	-	5 min. deviation only.
Military Rated Power	1150	3000	43.9	15,000	A. R.	104	359	330	5 min. Deviation only.
Normal Rated Power	1,000	2,600	38.7	15,000	A. R.	86	342	321	Highest permissible continuous Power <i>read climb</i>
Max. Cruising (75%)	750	2,280	31.6	15,000	A. R.	51	303	287	
Desired Cruising (67%)	670	2,280	28.9	15,000	A. R.	46	289	270	
					A. L.	42			
Desired Cruising (60%)	600	2,190	27.1	15,000	A. R.	41	274	255	
					A. L.	37.5			
Cruise for Minimum	375	1950	24.6	S/Level	A. L.	23	183	172	
	430	1950	24.6	5,000	A. L.	26			
Specific Fuel Flow	475	1950	24.6	10,000	A. L.	29	200	191	
	520	1950	24.6	15,000	A. L.	32	228	214	
	420	1950	Full Throttle	20,000	A. L.	32	254	238	
					A. L.	26	233	220	

NOTES : (a) To obtain maximum engine efficiency, manifold pressures shown against R.P.M. must not be exceeded when operating under normal conditions.

(b) Where 90 Octane fuel has to be used in the case of emergency, the following R.P.M. and Man. Pressures must not be exceeded and fuel tanks drained and refilled with 100 Octane immediately after conclusion of flight:-

Take off	2650	R. P. M.	37.0"	Hg. Man. Pressure
Climb	2450	"	30.0"	" " "
Cruise	2300	"	26.5"	" " "

KITTYHAWK

POWER TABLE - ALLISON V1710-39 ENGINE

100 OCTANE FUEL

Operating Condition	B.H.P.	R.P.M.	Man. press.	Altitude Density	Mixture Control	Fuel Flow G.P.H.	Remarks.
Take off	1150	2800	46.2	8/Level	A.R.	104	5 minutes duration only.
Military rated power	1150	3000	43.9	15,000	A.R.	104	5 minutes duration only.
Normal rated power	1000	2600	38.7	15,000	A.R.	86	Highest permissible continuous power and climb.
Maximum cruising (75%)	750	2280	31.6	15,000	A.R.	51	
Desired cruising (67%)	670	2280	28.9	15,000	A.R. A.L.	46 42	
Desired cruising (60%)	600	2190	27.1	15,000	A.R. A.L.	41 37.5	
Cruise for minimum specific fuel flow	375 430 475	1950 1950 1950	24.6 24.6 24.6	8/Level 5,000 10,000	A.L. A.L. A.L.	23 26 29	
	520 420	1950 1950	24.6 Full throttle	15,000 20,000	A.L. A.L.	32 26	

NOTES: (a) To obtain maximum engine efficiency, manifold pressures shown against R.P.M. must not be exceeded when operating under normal conditions.

(b) Where 90 octane fuel has to be used in the case of emergency, the following R.P.M. and Man. Pressures must not be exceeded and fuel tanks drained and refilled with 100 octane immediately after conclusion of flight:-

Take off	2650 R.P.M.	37.0" Hg.	Man. Pressure
Climb	2450 "	30.0" "	" "
Cruise	2300 "	26.5" "	" "

ENCLOSURE 4B.

FOR OFFICIAL USE ONLY

AIR FORCE HEAD-QUARTERS

AMENDMENT LIST NO. 1 TO ALLISON INSTRUCTIONS
AMENDMENT LIST NO. 1 TO KITTYHAWK INSTRUCTIONS

Allison Instruction No. 2 and Kittyhawk Instruction No. 2 are amended as follows:-

Remove and destroy existing "Power Table" and insert new Power Table (Issue 2) attached.

Reference: File R.A.A.F. 150/4/1847.

Date of Issue: 3rd July, 1942.

MITSUBISHI POWER TABLE - ALLISON V-1710-P3R (-89) (Issue 2) FUEL - 100 OCLINE.

OPERATING CONDITION	B.H.P.	R.P.M.	MANIFOLD PRESSURE	CRITICAL ALTITUDE	MIXTURE CONTROL	FUEL CONS. G.P.H.	REMARKS
Take off	1,180	2,300	46.2		Auto Rich	104	5 minute duration only.
Military rated power	1,130	2,000	44.6	10,700	Auto Rich	110	5 minute duration only.
Normal rated power	1,000	2,500	33.5	9,900	Auto Rich	37½	Highest permissible continuous power and climb.
Maximum cruising (75%)	750	2,280	30.8	9,900	Auto Rich	54	
Desired cruising (6%)	675	2,200	26.2	9,900	Auto Rich	46½	
					Auto Lean	41½	
Desired cruising (60%)	600	2,190	26.0	9,900	Auto Rich	41½	
					Auto Lean	37½	
Cruise for minimum specific fuel consumption	575	1,950	22.5	Sea Level	Auto Lean	33	
	490	1,950	22.5	5,000	Auto Lean	26½	
	460	1,950	22.5	10,000	Auto Lean	29	
	490	1,950	22.5	15,000	Auto Lean	31	
	450	1,950	P.T.	20,000	Auto Lean	28	

NOTES: (a) To obtain maximum engine efficiency, manifold pressures shown against r.p.m. must not be exceeded when operating under normal conditions.

(b) Where 90 octane fuel has to be used in the case of emergency, the following r.p.m. and manifold pressures must not be exceeded. Tanks must be drained before refuelling with 100 octane:-

Take off 2600 r.p.m. - 37" Hg. Man. Press.
 Climb 2450 " - 30" Hg. " "
 Cruise 2300 " - 26.5" Hg. Man. Press.

(A.L.1, July, 1942)